Notice No.4

Rules and Regulations for the Classification of Naval Ships, January 2021

The status of this Rule set is amended as shown and is now to be read in conjunction with this and prior Notices. Any corrigenda included in the Notice are effective immediately.

Please note that corrigenda amends to paragraphs, Tables and Figures are not shown in their entirety.

Issue date: June 2021

Amendments to	Effective date	IACS/IMO implementation (if applicable)
Volume 1, Part 1, Chapter 2, Section 3	Corrigenda	N/A
Volume 1, Part 3, Chapter 3, Section 2	Corrigendum	N/A
Volume 1, Part 3, Chapter 5, Sections 10 & 11	Corrigendum	N/A
Volume 1, Part 4, Chapter 2, Section 3	Corrigendum	N/A
Volume 1, Part 5, Chapter 3, Section 4	Corrigenda	N/A
Volume 1, Part 5, Chapter 4, Section 3	Corrigendum	N/A
Volume 1, Part 6, Chapter 2, Sections 3 & 4	Corrigenda	N/A
Volume 1, Part 6, Chapter 3, Sections 3, 14 & 15	Corrigenda	N/A
Volume 1, Part 6, Chapter 4, Section 2	Corrigenda	N/A
Volume1, Part 7, Chapter 3, Section 2	Corrigendum	N/A
Volume 2, Part 1, Chapter 3, Section 14	Corrigendum	N/A
Volume 2, Part 2, Chapter 1, Section 3	Corrigenda	N/A
Volume 2, Part 4, Chapter 2, Section 2	Corrigendum	N/A
Volume 2, Part 9, Chapter 8, Section 5	Corrigendum	N/A

Volume 1, Part 1, Chapter 2 **Classification Regulations**

Section 3

Character of Classification and Class notations

- Ship type notations 3.5
- NS(SR) and NS(SSC) vessels 3.5.6

For vessels that are using either the NS(SR) or NS(SSC) ship type notations, the following requirements are to be complied with:

- Plans of systems within the scope of classification, as categorised in accordance with Vol 2, Pt 1, Ch 1, 3.1 Categories and not covered by the Rules for Special Service Craft Rules and Regulations for the Classification of Special Service Craft or Rules for Ships Rules and Regulations for the Classification of Ships, as appropriate, are to be submitted to LR for approval in accordance with the requirements of the respective requirements of these Rules. The following are examples of such systems:

 - Chilled water systems (see Vol 2, Pt 7, Ch 2 Ship Piping Systems Vol 2, Pt 7, Ch 5 Ship Type Piping Systems). High pressure sea-water systems (see Vol 2, Pt 7, Ch 2 Ship Piping Systems Vol 2, Pt 7, Ch 5 Ship Type Piping iii.
 - iv. High and low pressure compressed air systems (see Vol 2, Pt 7, Ch 2 Ship Piping Systems Vol 2, Pt 7, Ch 5 Ship Type Piping Systems).
 - Hydraulic power actuating systems (see Vol 2, Pt 7, Ch 2 Ship Piping Systems Vol 2, Pt 7, Ch 5 Ship Type Piping Systems).

Volume 1, Part 3, **Chapter 5**

Anchoring, Mooring, Towing, Berthing, Launching, Recovery and Docking

Section 10

Anchoring equipment in deep and unsheltered water

10.3 Anchor windlass and chain stopper

Table 5.10.1 Anchoring equipment for ships in unsheltered water with depth up to 120 m

Equipment Number EN ₁		High holding power stockless bower anchors		Stud link chain cable for bower anchors		er anchors
					Min. di	ameter
Equal to or greater than	Less than	Number	Mass per anchor (kg)	Length (m)	Special quality (Grade U2) (mm)	Extra special quality (Grade U3) (mm)
 8400 	8900	2	2800 28000	797,5	158	127

Section 11

Launch and recovery, berthing and dry-docking arrangements

11.4 **Dry-docking loads**

11.4.6 The following equation may be used to calculate the dry-docking load distribution, F-DL, between main transverse bulkheads acting on a keel block:

 $f_{bhd} = 2$ 0,5, for the keel blocks located adjacent to a main transverse bulkhead

Volume 1, Part 4, Chapter 2 **Military Load Specification**

- Section 3 Internal blast
- Quasi static pressure 3.5
- The QSP can be determined from the following: 3.5.3
- $= 2.25 (W_e M)^{0.72} \times 10^3 kN/m^2$

 $P_{\rm qs} = 2.25 \ (W_{\rm e} \ V/)^{0.72} \ {\rm x} 10^3 \ {\rm kN/m^2}$

Volume 1, Part 5, **Chapter 3 Local Design Loads**

- Section 4 Impact loads on external plating
- 4.2 Bottom impact pressure, IP bi
- The bottom impact pressure due to slamming, IPbi, is to be derived using the method given below. This method will produce impact pressures over the whole of the underwater plating region:

where

= slamming velocity, in m/s, and is given by $V_{
m bs}$

$$= \sqrt{V_{th} + 2m_1 IN(N_{s1})} \text{ for } N_{s1} \ge 1$$

$$\sqrt{{V_{\text{th}}}^2 + 2m_1 ln(N_{\text{s1}})}$$
 for $N_{\text{s1}} \ge 1$

 $= 0 \text{ for } N_{s1} < 1$

4.3 Bow flare and wave impact pressures, IPbf

- 4.3.1 This Section is applicable to:
- Bow flare region.
- Sides and undersides of sponsons. (b)
- Other parts of the side shell plating close to and above the design waterline that are expected to be subjected to wave impact

The bow flare wave impact pressure, wave impact pressure on sponsons and other parts of the side shell plating above the design waterline, IP bf, in kN/m², due to relative motion is to be taken as:

where

 V_{bf} = wave impact velocity, in m/s, and is given by

$$= \sqrt{V_{ERDf}^2 + 2m_1 IN(N_{Df})} \text{ for N}_{bf} > 1 \text{ for N}_{bf} \ge 1$$

$$\sqrt{V_{\text{thbf}}^2 + 2m_1 ln(N_{\text{bf}})} \text{ for } N_{\text{bf}} \ge 1$$

Volume 1, Part 5, Chapter 4 Global Design Loads

Section 3Global hull girder loads

3.3 Vertical wave bending moments

3.3.1 The minimum value of vertical wave bending moment, M_{W} at any position along the ship may be taken as follows: $M_{W} = F_{T}D_{T}M_{O}$ kNm

An area ratio value of 1,0 results in a sagging correction of factor -1,10.

- (a) FfH F_H is the hogging (positive) moment correction factor and is to be taken as
- 3.3.2 The area ratio factor, RA, for the combined stern and bow shape is to be derived as follows:
- 3.3.3 The bow flare area, A_{BF} A_{BF} , is illustrated in *Figure 4.3.1 Deviation of bow and stern flare areas* and may be derived as follows:

 $T_{C,U} = \frac{is}{s}$ a waterline taken $L_{-1}/2$ m above the design draught

 $T_{c,u} = T + L_f/2 \text{ m}$

where

b₀ = projection of Tc,u Tc,u waterline outboard of the design draught waterline at the FP, in metres, see Figure 4.3.1 Deviation of bow and stern flare areas

 b_1 = projection of $\mp_{C,U}$ $T_{C,U}$ waterline outboard of the design draught waterline at $0.9L_R$ from the AP, in metres

 b_2 = projection of $\mp_{C,U}$ $T_{C,U}$ waterline outboard of the design draught waterline at 0,8 L_R from the AP, in metres

3.3.4 The stern flare area, A_{SF} A_{SF}, is illustrated in *Figure 4.3.1 Deviation of bow and stern flare areas* and is to be derived as follows:

 $T_{C,L}$ = is a waterline taken $L_f/2$ m below the design draught

 $T_{C,L} = \frac{T - L_f - m^2}{T - L_f/2} T - L_f/2 m.$

3.3.8 The sagging correction factor, fis, in the vertical wave bending moment formulation in *Vol 1, Pt 5, Ch 4, 3.3 Vertical wave bending moments 3.3.1* may be derived by direct calculation methods. Appropriate direct calculation methods

3.8 Bow flare impact global loads

- 3.8.1 The requirements of this section are applicable to fast ships operating in the displacement mode that satisfy the following requirements:
- a) speed $V_{sp} > 17,5$ knots
- b) bow shape factor ψ > 0,15

Volume 1, Part 6, Chapter 2 **Design Tools**

Section 3 **Buckling**

3.3 Plate panel buckling requirements

Table 2.3.1 Plate panel buckling requirements

	able 2.0.1 Flate parter backing requirements				
	Stress field	Buckling interaction formula			
(c)	bi-axial compressive loads	for $A_R = 1.0$			
	·	for other aspect ratios, i.e. $A_R \neq 1,0$ when G is taken from Figure 2.3.3 Interaction limiting stress curves of G for plate panels			
		subject to bi-axial compression , see Table 2.3.2(c)			

Figure 2.3.3 Interaction limiting stress curves of G for plate panels subject to bi-axial compression, see Table 2.3.2(c)

Table 2.3.3 Buckling stress of secondary stiffeners

 σ_{ep} = elastic eritical buckling stress, σ_{e} , in N/mm², of the supporting plate derived from Table 2.4.1 First mode of vibration constant Ki Table 2.3.2 Buckling stress of plate panels, (a) i)

Section 4 Vibration control

4.4 Natural frequency of plate and stiffener combination

4.4.1 The natural frequency of a plate stiffener of constant cross-section in air is given by the following:

$$f_{\overline{air}} = \frac{k_{t}}{20\pi l_{bz}} \sqrt{\frac{El}{m\left(1 + \frac{\pi_{z}El}{104 l_{bz}GA}\right)}} HZ$$

$$f_{\text{air}} = \frac{k_{\text{i}}}{20\pi l_{\text{b}}^2} \sqrt{\frac{EI}{m\left(1 + \frac{\pi^2 EI}{10^4 l_{\text{B}}^2 \text{GA}}\right)}} \text{Hz}$$

Volume 1, Part 6, **Chapter 3 Scantling Determination**

Section 3 **NS1** scantling determination

3.10 Shell envelope framing

	Symbols		
F s-	Fs- = fatigue factor for side longitudinals for built symmetric sections, flat bars, bulbs and T		
	bars:		
	= 1,05 at keel, 1,1 at 7, 1,0 at 1,67 and above		
	For angle bars:		
	$=0.5\left(1+\frac{1.1}{k_{\pi}}\right)$ at keel		
	$=\frac{\frac{1}{1}}{k_{\mathcal{E}}}\frac{dt}{2}$		
	= 1,0 at 1,6 T and abovebuilt asymmetric sections will be specially considered.		
	Intermediate values by linear interpolation		

1

F s	= fatigue factor for side longitudinals
	For built symmetric sections, flat bars, bulbs and <i>T</i> bars:
	= 1,05 at keel, 1,1 at T , 1,0 at 1,6 T and above
	For angle bars:
	$=0.5\left(1+\frac{1.1}{k_{\rm S}}\right)$ at keel
	$=\frac{1.1}{k_S} \text{ at } \frac{D}{2}$
	= 1,0 at 1,6 T and above
1	Intermediate values by linear interpolation
1	Built asymmetric sections will be specially considered.

Table 3.3.4 Shell envelope primary structure

Table City City City City City City City City		
Item and location	Modulus, in cm ³	
Transverse framing system:		
(4) ∠ Side stringers in dry spaces		
≠ (5) Side stringers in deep tanks	$Z = 9.4K_s$ S $h 4l_e^2$ or as $\frac{(5)}{(4)}$ above, whichever is the greater	
(6) Web frames in dry spaces above 1,6 <i>T</i> (see Note 2)	$Z = C_3 k_s STHD$ $Z = C_3 k_s STH\sqrt{D}$	

Table 3.3.6 Deck plating

Location	Minimum thickness, in mm, see also Vol 1, Pt 6, Ch 3, 2.2 Corrosion margin		
	Longitudinal framing	Transverse framing	
(3) Lower decks (a) effective (continuous) (b) non effective	$t = 0.011s_1\sqrt{k_s}t = 0.009s_1\sqrt{k_s}$ $t = 0.009s_1\sqrt{k_s}$		
(4) Strength deck (a) forward of 0,925LR And aft of 0,075LR (b) Lower decks	$t = (5.0 + 0.018L_{\rm R}) \sqrt{\frac{k_{\rm S}S_1}{s_{\rm b}}} t = \frac{0.009s_{\rm I}\sqrt{k_{\rm S}}}{t}$ $t = 0.009s_{\rm I}\sqrt{k_{\rm S}}$		

■ Section 14

Strengthening for bottom slamming

14.2 Strengthening of bottom forward

Table 3.14.1 Additional strengthening of bottom forward

Item	Requirements	
(2) Bottom longitudinals – other than flat bars	$\frac{22000}{2000}$	
	$Z \ge 6.8 \times 10^{-6} h_{\rm s} s k_{\rm s} \left[(17.5 l_{\rm s})^2 - (0.01 l_{\rm s})^2 + d_{\rm W} c \left(S - \frac{S}{2000} \right) \right] {\rm cm}^3$	

Section 15

Strengthening for wave impact loads above waterline

15.3 Strengthening against wave impact loads

Table 3.15.1 Buckling procedure for primary member web plating and web stiffener

Table 5.15.1 Buckling procedure for primary member web plating and web stillener				
Steps	Members			
	Primary member web plating	Primary member web stiffener		
Determination of the elastic critical buckling stress, σ_e , in compression, N/mm ² (kgf/mm ²)	$\sigma_{e} = \frac{9,87 E I_{W}}{1_{W} A_{W}^{2}}$ $\sigma_{e} = \frac{9,87 E I_{W}}{1_{W}^{2} A_{W}}$	$\sigma_{e} = \frac{_{9,87EI_{S}}}{_{1_{S}A_{S}}^{2}} \qquad \qquad \sigma_{e} = \frac{_{9,87EI_{S}}}{_{1_{S}^{2}A_{S}}}$		
Determination of the corrected critical buckling stress, σ _{cr} , in compression, N/mm ² (kgf/mm ²)	$\sigma_{\rm cr} = \sigma_{\rm g} \left(1 - \frac{\sigma_{\rm g}}{\sigma_{\rm e}} \right) \ \sigma_{\rm cr} = \sigma_0 \left(1 - \frac{\sigma_0}{4\sigma_{\rm e}} \right)$	where $\sigma_{\rm e} > \frac{\sigma_{\rm o}}{2}$		

Volume 1, Part 6, Chapter 4 Hull Girder Strength

- Section 2Hull girder strength
- 2.3 Shear strength

Table 4.2.2 k i factors

k _i factors
Member 3
$k_3 = -0.01 \frac{A_3}{A_4} + 0.25$
$k_3 = 0.01 \frac{A_3}{A_4} + 0.25$
Member 4
$k_4 = 0.01 \frac{A_3}{A_4} + 0.25$
$k_4 = -0.01 \frac{A_3}{A_4} + 0.25$

2.5 Super structures global strength

2.5.3 The design stress due to hull girder bending, σ_{hg} , in the uppermost effective tier at side may be derived according to the following formula:

$$\sigma_{\overline{h}\overline{g}} = \frac{\eta_s M_R}{1000Z_s} \text{ N/mm2} \qquad \qquad \sigma_{hg} = \frac{\eta_s M_R}{1000Z_s} \text{ N/mm}^2$$

where

M-R = hull girder bending moment at amidships due to sagging as determined in, Vol 1, Pt 5, Ch 4, 5 Residual strength hull girder loads, vol 1, Pt 5, Ch 4, 3 Global hull girder loads, in kNm

Volume 1, Part 7, Chapter 3 Total Load Assessment, TLA

- Section 2Structural resistance
- 2.2 Stresses in plating
- 2.2.2 The bending stress in a plate panel between stiffeners due to a uniform lateral pressure is to be calculated as follows:

3

$$\sigma_{\!\scriptscriptstyle p} = p \left(\frac{22, 4_s \gamma \beta}{100 t_p} \right) N/m m^2$$

$$\sigma_{\rm b} = p \left(\frac{22,4s\gamma\beta}{1000t_p} \right)^2 \text{ N/mm}^2$$

Volume 2, Part 1, Chapter 3

Requirements for Design, Construction, Installation and Sea Trials of Engineering Systems

■ Section 14
Thrusters

14.1 Design and construction

14.1.1 For details of design and construction requirements, see Vol 2, Pt 4, Ch 3 Water Jet Systems Vol 2, Pt 4, Ch 3 Thrusters.

Volume 2, Part 2 Chapter 1 Reciprocating Internal Combustion Engines

Section 3Crankshaft Design

3.2 Scope

3.2.9 Further information and guidance on crankshaft design is provided in the LR's Guidance Notes for the Calculation of Stress Concentration Factors, Fatigue Enhancement Methods and Evaluation of Fatigue Tests for Crankshafts.

3.3 Information to be submitted

- 3.3.2 The following information is also required for appraisal of the crankshaft (not contained in Form 2073):
- every surface treatment affecting fillets or oil holes shall be specified so as to enable calculation according to Chapter 2 3 of the LR Guidance Notes for Grankshaft SCF Calculation using Finite Element Method the Calculation of Stress Concentration Factors, Fatigue Enhancement Methods and Evaluation of Fatigue Tests for Crankshafts;
 - o this is to include Grankshaft crankshaft fatigue enhancement factors K_4 K_1 and K_2 K_2 where applicable.

3.4 Symbols

- 3.4.1 For the purposes of this Chapter the following symbols apply, see also;
- D_A = the outside diameter of we or twice the minimum distance between eentre line centreline of journals and outer contour of web, whichever is less, in mm

 $W_{\text{eqw}} = \frac{\text{xection}}{\text{section}}$ section modulus related to cross-section of web, in mm³

- y = distance between the adjacent generating lines of journal and pin, in mm Note for $y \ge 0.5 D_s$ where,w. Where y is less than $0.1 D_s$, special consideration is to be given to the effect of the stress due to the shrink fit on the fatigue strength at the crankpin fillet.
- σ_y = equivalent alternating stress for crankpin fillet, journal fillet or outlet of crankpin oil bore as applicable, in $\frac{N}{mm^3}$ $\frac{N}{mm^2}$

3.5 Calculation of alternating stresses due to bending moments and radial forces – assumptions

Figure 1.3.9 Bending moment and shear force for in-line engine crankthrows has been replaced with the below figure;

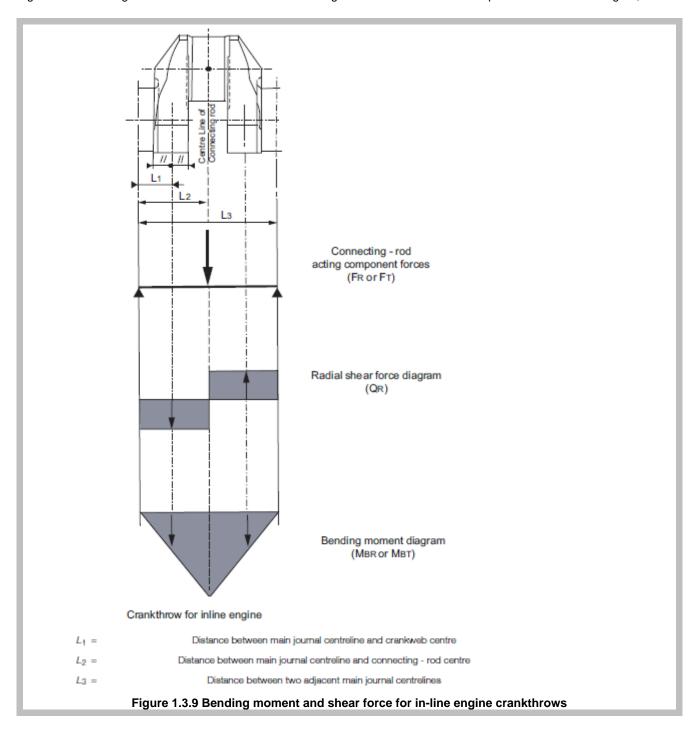
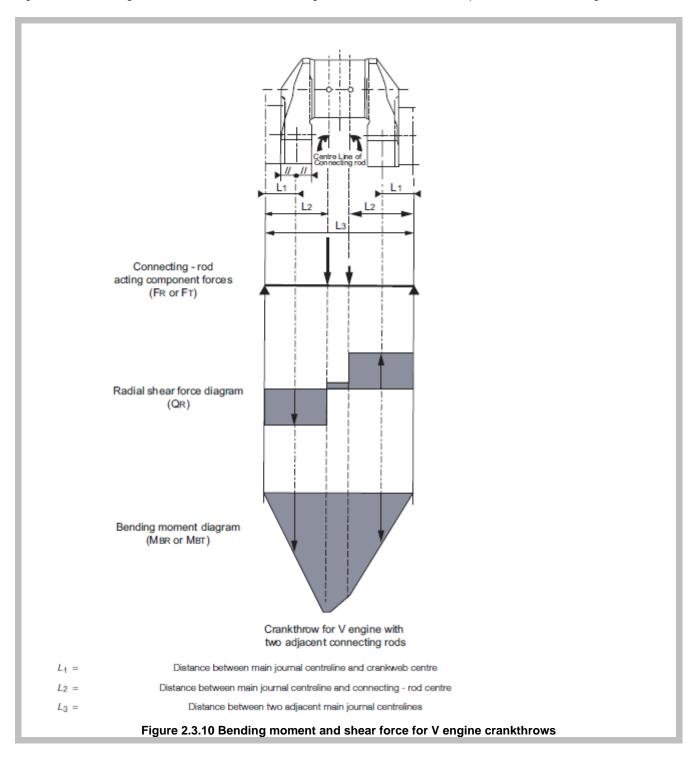


Figure 1.3.10 Bending moment and shear force for V engine crankthrows has been replaced with the below figure;



3.5.4 The two relevant bending moments for bending acting on the outlet of crankpin oil bores are taken in the crankpin cross-section through the oil bore. See Figure 1.3.9 Bending moment and shear force for in-line engine crankthrows and Figure 1.3.10 Bending moment and shear force for V engine crankthrows. MBRQ is the bending moment of the radial component of the connecting-rod force and MBTQ is the bending moment of the tangential component of the connecting-rod force. The alternating stresses due to these bending moments are to be related to the cross-sectional area of the axially bored crankpin. Mean bending stresses are neglected

The two relevant bending moments for bending acting on the outlet of crankpin oil bores are taken in the crankpin cross-section through the oil bore. See;

- Figure 1.3.5 Crankpin section through the oil bore,
- Figure 1.3.9 Bending moment and shear force for in-line engine crankthrows, and
- Figure 1.3.10 Bending moment and shear force for V engine crankthrows.

 $M_{\rm BRO}$ is the bending moment of the radial component of the connecting-rod force and $M_{\rm BTO}$ is the bending moment of the tangential component of the connecting-rod force. The alternating stresses due to these bending moments are to be related to the cross-sectional area of the axially bored crankpin. Mean bending stresses are neglected.

3.6 Calculation of bending stresses

- 3.6.1 The radial and tangential forces due to gas and inertia loads acting on upon the crankpin at each connecting-rod position are to be calculated over one working cycle. Using the forces calculated over one working cycle and taking into account of the distance from the main bearing midpoint, the time curve of the bending moments, M_{BRF} , M_{BRO} and M_{BTO} , and radial forces, Q_{RF} , as defined in $Vol\ 2$, $Pt\ 2$, $Ch\ 1$ Calculation of alternating stresses due to bending moments and radial forces assumptions 3.5.2 3.5.3 and $Vol\ 2$, $Pt\ 2$, $Ch\ 1$ Calculation of alternating stresses due to bending moments and radial forces assumptions 3.5.3 3.5.4 are then calculated.
- 3.6.6 Nominal alternating bending and compressive stresses in a web cross-section are calculated as follows:

$$\sigma_{\rm BFN} = \pm \frac{M_{\rm BRFN}}{W_{\rm eqw}} 10^3 \ K_{\rm e} \ {\rm N/mm^2}$$

$$\sigma_{QFN} = \pm \frac{Q_{RFN}}{F} K_e \text{ N/mm}^2$$

where

$$\begin{aligned} & \mathcal{M}_{\text{BRFN}} = \pm \frac{1}{2} (X_{\text{BRF Max}} - X_{\text{BRF Min}}) \text{Nm} \\ & \mathcal{Q}_{\text{RFN}} = \pm \frac{1}{2} (Q_{\text{RF Max}} - Q_{\text{RF Min}}) \text{Nm} \\ & \mathcal{Q}_{\text{RFN}} = \pm \frac{1}{2} (Q_{\text{RF Max}} - Q_{\text{RF Min}}) \text{Nm} \end{aligned}$$

3.6.7 Nominal alternating bending stress in the outlet of the crankpin oil bore is calculated as follows:

$$\sigma_{\text{BON}} = \pm \frac{M_{\text{BON}}}{W_{\text{e}}} 10^3 \, \text{N/mm}^2$$

where

 M_{BON} is taken as the ½ range value $M_{BON} = \pm \frac{1}{2} (M_{BOmax} - M_{BOMin})$ Nm

3.7 Calculation of torsional stresses

- 3.7.1 The nominal alternating torsional stress, $\tau_{\overline{n}}$ τ_{N} , is to be taken into consideration. The value is to be derived from forced-damped vibration calculations of the complete dynamic system. Alternative methods will be given consideration. The engine designer is to advise the maximum level of alternating vibratory stress that is permitted (τ_{a}) .
- 3.7.2 ## In or ## In (as applicable) is to be applied as a limiting value for the torsional vibration assessment required by Vol 2, Pt 5, Ch 1 Torsional vibration.
- 3.7.3 Nominal alternating torsional stress is calculated as follows:

$$au_{\overline{\mathrm{H}}} \ au_{\mathrm{N}} = rac{\mathit{M}_{\mathrm{TN}}}{\mathit{W}_{\mathrm{P}}} \, 10^3 \ \mathrm{N/mm}^2$$

where

$$\begin{split} W_{p} &= \frac{\pi}{46} \binom{D^{4} - D_{BH}^{4}}{D} \text{ mm}^{3} \text{ for the crankpin, or } W_{p} = \frac{\pi}{46} \binom{D_{G}^{4} - D_{BG}^{4}}{D_{c}} \text{ mm}^{3} \text{ for the jorunal} \\ W_{p} &= \frac{\pi}{16} \binom{D^{4} - D_{BH}^{4}}{D} \text{ mm}^{3} \text{ for the crankpin, or } W_{p} = \frac{\pi}{16} \binom{D_{G}^{4} - D_{BG}^{4}}{D_{c}} \text{ mm}^{3} \text{ for the journal} \end{split}$$

3.8 Stress concentration factors

- 3.8.3 Where the geometry of the crankshaft is outside the boundaries (see Table 1.3.2 Crankshaft variable boundaries for analytical SCF calculation) of the analytical SCG the calculation method detailed in Chapter 1 and Chapter 4 of the LR Guidance Notes for Crankshaft SCG Calculation using Finite Element Method Calculation of Stress Concentration Factors, Fatigue Enhancement Methods and Evaluation of Fatigue Tests for Crankshafts may be undertaken.
- 3.8.5 Chapters1 and 3 4 of the LR Guidance Notes for Crankshaft SCG Calculation using Finite Element Method Calculation of Stress Concentration Factors, Fatigue Enhancement Methods and Evaluation of Fatigue Tests for Crankshafts describe how finite element (FE) analyses can be used for the calculation of the SCF. Care needs to be taken to avoid mixing equivalent (von Mises) stresses and principal stresses.
- 3.8.6 Crankpin SCF are calculated as follows:
- a) Bending

 $\alpha_B = 2,6914f(s,w).f(w).f(b).f(r).f(d_G).f(d_H).f(recess)$

$$\alpha_{\rm B} = 2,6914 \cdot f(s,w) \cdot f(w) \cdot f(b) \cdot f(r) \cdot f(d_{\rm G}) \cdot f(d_{\rm H}) \cdot f({\rm recess})$$

b) Torsion

$$\alpha_{T} = 0.8f(r,s).f(b).f(w)$$

$$\alpha_{T} = 0.8f \cdot (r,s) \cdot f(b) \cdot f(w)$$

$$f(b) = 7,8955 - 10,654b + 5,3482b^2 - 0,857b^3f(w) = w^{-0,145}$$

 $f(w) = w^{-0,145}$

- 3.8.7 Journal fillet SCF are calculated as follows (not applicable to semi-built) crankshafts):
- Bending a)

$$\beta_{\rm B} = 2,7146 f_{\rm B}(s,w).f_{\rm B}(w).f_{\rm B}(b).f_{\rm B}(r).f_{\rm B}(d_{\rm G}).f_{\rm B}(d_{\rm H}).f(recess)$$

$$\beta_{B} = 2,7146 \cdot f_{B}(s,w) \cdot f_{B}(w) \cdot f_{B}(b) \cdot f_{B}(r) \cdot f_{B}(d_{G}) \cdot f_{B}(d_{H}) \cdot f(\text{recess})$$

Compression due to the radial force:

$$\beta_Q = 3.0128 f_Q(s).f_Q(w).f_Q(b).f_Q(r).f_Q(d_H).f(recess)$$

$$\beta_0 = 3.0128 \cdot f_0(s) \cdot f_0(w) \cdot f_0(b) \cdot f_0(r) \cdot f_0(d_H) \cdot f(recess)$$

3.9 Additional bending stress

In addition to the alternating bending stresses in fillets (seeVol 2, Pt 2, Ch 1, 3.6 Calculation of bending stresses 3.6.8) further bending stresses due to misalignment and bedplate deformation as well as due to axial and bending vibrations are to be considered by applying α_{add} σ_{add} as given by Table 1.3.3 Additional bending stresses.

3.10 **Equivalent alternating stress**

- Equivalent alternating stress, σ_v , is defined as:
- For the crankpin fillet:

$$\sigma_{V} = \pm \sqrt{(\sigma_{BG} + \sigma_{add})^2 + 3\tau_{G}^2 N/mm^2}$$

$$\sigma_{\text{v}} = \pm \sqrt{(\sigma_{\text{BH}} + \sigma_{\text{add}})^2 + 3\tau_{\text{H}}^2} \text{ N/mm}^3$$

For the journal fillet:

(b) For the journal fillet:

$$\sigma_{V} = \pm \sqrt{(\sigma_{BH} + \sigma_{add})^{2} + 3\tau_{H}^{2} \text{N/mm}^{3}}$$

$$\sigma_{V} = \pm \sqrt{(\sigma_{BG} + \sigma_{add})^2 + 3\tau_{G}^2} \, \text{N/mm}^2$$

3.11 **Fatigue strength**

The fatigue strength is to be understood as the value of equivalent alternating stress (Von von Mises) which a crankshaft can permanently withstand at the most highly stressed points. The fatigue strength can be evaluated by means of the following formulae:

$$\begin{split} &\sigma_{\rm DW} = \pm K(0.42\sigma_{\rm B} + 39.3) \left[0.264 + 1.073D^{-0.2} + \frac{785 - \sigma_{\rm B}}{4900} + \frac{196}{\sigma_{\rm E}} \sqrt{\frac{1}{R_{\rm K}}} \right] {\rm N/mm^2} \\ &\sigma_{\rm DW} = \pm K(0.42\sigma_{\rm B} + 39.3) \left[0.264 + 1.073D^{-0.2} + \frac{785 - \sigma_{\rm B}}{4900} + \frac{196}{\sigma_{\rm B}} \sqrt{\frac{1}{R_{\rm X}}} \right] {\rm N/mm^2} \end{split}$$

b) Related to the journal diameter:

$$\begin{split} & \sigma_{\rm DW} = \pm K (0.42\sigma_{\rm B} + 39.3) \left[0.264 + 1.073 D_{\rm G}^{-0.2} + \frac{785 - \sigma_{\rm B}}{4900} + \frac{196}{\sigma_{\rm E}} \sqrt{\frac{1}{R_{\rm G}}} \right] {\rm N/mm^2} \\ & \sigma_{\rm DW} = \pm K (0.42\sigma_{\rm B} + 39.3) \left[0.264 + 1.073 D^{-0.2} + \frac{785 - \sigma_{\rm B}}{4900} + \frac{196}{\sigma_{\rm B}} \sqrt{\frac{1}{R_{\rm X}}} \right] {\rm N/mm^2} \end{split}$$

A value for K_2 will be assigned upon application by the engine designers. Full details of the process, together with the results of full scale fatigue tests will be required to be submitted for consideration. See Chapter 2 of the LR Guidance note - Guidance for the evaluation of Crankshaft Fatigue Tests Notes for the Calculation of Stress Concentration Factors, Fatigue Enhancement Methods and Evaluation of Fatigue Tests for Crankshafts.

- Fatigue strength calculations or alternatively, fatigue test results determined by experiment based either on full size crankthrow (or crankshaft), or on specimens taken from a full size crankthrow, may be required to demonstrate acceptability. The experimental procedure for fatigue evaluation of specimens and fatigue strength of crankshaft assessment are to be submitted for approval by LR. The procedure is to include as a minimum: method, type of specimens, number of specimens (or crankthrows), number of tests, survival probability, and confidence number. See also Chapter 2 of the LR Guidance for the Evaluation of Crankshaft Fatigue Tests Notes for the Calculation of Stress Concentration Factors, Fatigue Enhancement Methods and Evaluation of Fatigue Tests for Crankshafts. Alternatively, the following values may be taken (surface hardened zone to include fillet radii):
- 3.11.5 Only surface treatment processes approved by LR are permitted. Guidance for calculation of surface treated fillets and oil bore outlets is presented in Chapter 2 3 of the LR Guidance Notes for Crankshaft SCF Calculation using Finite Element Method the Calculation of Stress Concentration Factors, Fatigue Enhancement Methods and Evaluation of Fatigue Tests for Crankshafts.

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3.13 Shrink fit of semi-built crankshafts

3.13.5 The actual oversize Z of the shrink fit must be within the limits Z_{min} and Z_{max} calculated in accordance $Vol\ 2$, $Pt\ 2$, $Ch\ 1$, 3.13 Shrink fit of semi-built crankshafts 3.13.7 3.13.6 and $Vol\ 2$, $Pt\ 2$, $Ch\ 1$, 3.13 Shrink fit of semi-built crankshafts 3.13.7. When $Vol\ 2$ $Pt\ 2$, $Ch\ 1$, 3.13 Shrink fit of semi-built crankshafts 3.13.4 cannot be complied with, then the calculated values of Z_{min} and Z_{max} are not applicable due to multizone-plasticity problems. In such cases Z_{min} and Z_{max} are to be established from FEM calculations.

3.13.6 The minimum required diametral Interference interference is to be taken as the greater of:

$$Z_{\min} \ge \frac{\sigma_{SW} D_S}{E_m}$$
 mm

and

$$Z_{\min} \geq \frac{4000}{\mu \pi} \frac{S_{\rm R} \, M_{\max}}{E_{\rm m} D_{\rm S} L_{\rm S}} \frac{1 - {Q_{\rm A}}^2 {Q_{\rm S}}^2}{\left(1 - {Q_{\rm A}}^2\right) \left(1 - {Q_{\rm S}}^2\right)} \ \ {\rm mm}$$

where

$$Q_S = \frac{D_{BG}}{D_S}$$
 shaft ratio, $Q_S = \frac{D_{BG}}{D_S}$

Volume 2, Part 4 Chapter 2 Water Jet Systems

■ Section 2

General requirements

2.1 Water jet arrangement

2.1.1 In general, for a ship to be assigned an unrestricted service notation, a minimum of two water jet systems is to be provided where these form the sole means of propulsion. For ships where a single water jet system is the sole means of propulsion or steering, a detailed engineering and safety justification is to be evaluated by LR, see Vol 2, Pt 4, Ch 2, 2.3 Calculations and information 2.3.22 2.3.23. This evaluation process will include a Risk Assessment (RA) in accordance with Vol 2, Pt 1, Ch 3, 18 Risk Assessment (RA), to verify that sufficient levels of redundancy and monitoring are incorporated in the water jet unit's support systems and operating equipment.

Volume 2, Part 9 Chapter 8 Programmable Electronic Systems

Section 5

Programmable electronic systems (PES)

5.1 General requirements

5.1.21 Software lifecycle activities, e.g. design, development, supply and maintenance, are to be carried out in accordance with an acceptable quality management system. Project specific software quality plans are to be submitted. These are to demonstrate that the provisions of ISO/IEC 90003:2014 Software engineering – Guidelines for the application of ISO 9001:2008 2015 to computer software or an acceptable International, National or naval standard, are incorporated. The plans are to define responsibilities for the lifecycle activities, including verification, validation, module testing and integration with other components or systems.

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